## **DESIGN OF BEAM**

# **LATERALLY SUPPORTED:**

STEP 1: FIND OUT ULTIMATE LOAD ON BEAM.

Factored Ultimate Load (Factored Load) w = 1.5 × Working Load

STEP 2: FIND OUT MAXIMUM BENDING MOMENT (M) AND SHEAR FORCE (V) ON BEAM.

STEP 3: CALCULATE PLASTIC SECTION MODULUS REQURIED FOR TRIAL SECTION.

$$Z_{P(required)} = \frac{M\gamma_0}{f_v}$$

STEP 4: SELECT SUITABLE SECTION BASED ON  $Z_p$  FROM IS: 800: 2007, PAGE NO. 138, 139. WRITE DOWN SE TIONAL PROPERTIES.

#### STEP 5: SECTION CLASSIFICATION.

a. Find out value of  $b/t_f$  and  $d/t_w$ . (refer Figure. 2, Page no. 19, IS 800: 2007 to find b and d)  $t_f$  = thickness of flange  $t_w$  = thickness of web.

b. Refer Table 2, Page no. 18, IS 800: 2007 and classify the section semi-compact, compact, plastic or slender.

**STEP 6: CHECK FOR SHEAR.** (Clause no. 8.4.1., Page no. 59, IS 800: 2007)

a. Find out design shear strength V<sub>d</sub>.

$$V_d = \frac{f_y}{\sqrt{3\gamma_{mo}}} ht_w$$

### b. Beam is checked for high / low shear case

 $V \le 0.6 V_d$  low shear case

 $V > 0.6 V_d$  high shear case

## STEP 6: CHECK FOR BENDING.

**a. For low shear Case** (Clause no. 8.2.1.2, Page no. 53, IS 800: 2007)

$$M_d > M$$

M<sub>d</sub> = Design Bending Strength

M = Bending Moment

$$M = \beta \ Z \ \underline{f_{y}} \le 1 \mathbb{Z}$$
 (for simply supported beam)

d b 
$$_{p}$$
  $\gamma_{mo}$   $_{e}$   $\gamma_{mo}$   $\leq 1.5Z_{e} \frac{f_{\gamma}}{\gamma_{mo}}$  (for cantilever beam)

 $\beta_b$  = 1 for plastic and compact sections.

=  $Z_e/Z_p$  for semi compact sections.

Z<sub>e</sub> = Elastic section Modulus

Z<sub>p</sub> = Plastic section Modulus

**b. For High shear Case** (Clause no. 8.2.1.3, Page no. 53, IS 800: 2007)

Refer Clause no. 8.2.1.3, Page no. 53, IS 800: 2007. Generally low shear case is preferred.

STEP 7: CHECK FOR WEB BUCKLING AT SUPPORT (Clause no. 8.7.3.1, Page no. 67, IS 800: 2007)

a. Capacity of section =  $A_b f_{cd} > V$ 

b.  $A_b = (b_1 + n_1) t_w$  when load is at support

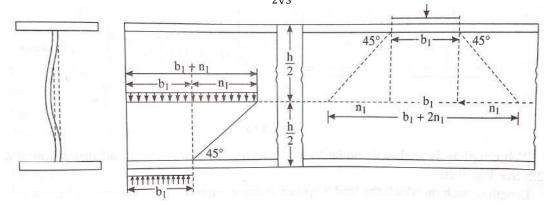
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 $A_b = (b_1 + 2n_1) t_w$  when load is not at support

Where,  $b_1$ = stiff bearing length of load = assume between 0 to 100mm  $n_1$  = for 45° dispersion consider h/2

d. Find out  $F_{cd}$  = Design Compressive Stress considering class c and  $f_y$  = 250 MPa. Slenderness ratio =  $\frac{kl}{l} = \frac{0.7d}{l}$ 

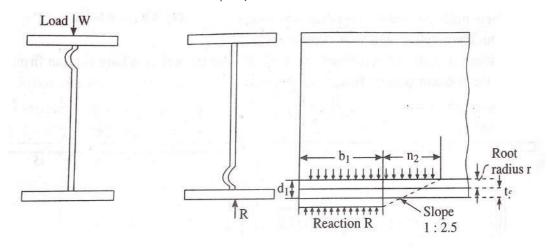
D = depth of the web between the flanges r = least radius of gyration of the section =  $\frac{t_w}{2\sqrt{3}}$ 



**STEP 7: CHECK FOR WEB CRIPPLING** (Clause no. 8.7.4, Page no. 67, IS 800: 2007)  $(h_1+n_2)t_3...f_3...$ 

Design crippling strength 
$$F_w = \frac{(b_1 + n_2)t_w f_{yw}}{\gamma_{mo}} > V$$

Where, 
$$b_1$$
 = stiff bearing length = 0 to 100 mm  
 $n_2$  = 2.5 (  $t_f$ + $r_1$ )  
 $f_{yw}$  = yield stress of web



# **STEP 8: CHECK FOR DEFLECTION**

a. Actual deflection for simply supported

$$\delta_{max} = \frac{5 \ wl^2}{384 \ EI}$$

b. Permissible deflection = Span/300 (table 6, Page no. 31, IS 800: 2007)