## 5.4 Solved Problems Boundary Layer Thickness

PROBLEM 1: The velocity distribution in the boundary layer is given by

$$\frac{u}{U} = \frac{y}{\sigma}$$

where u is the velocity y from the plate and u=U at , y = $\delta$  , $\delta$  being boundary layer thickness.Find

- i. The displacement thickness
- ii. The momentum thickness
- iii. The energy thickness and
- iv. The value of  $\delta */ \theta$ .

## **Solution:**

Velocity distribution:

$$\frac{u}{U} = \frac{y}{\sigma}$$

(i) The displacement thickness  $\delta^*$ 

$$\delta^* = \int_0^{\delta} \left( 1 - \frac{u}{U} \right) dy$$
$$= \int_0^{\delta} \left( 1 - \frac{y}{\delta} \right) dy$$

$$= \left[ y - \frac{y^2}{2\delta} \right]_0^{\delta}$$

$$= \left( \delta - \frac{\delta^2}{2\delta} \right) = \delta - \frac{\delta}{2}$$

$$= \frac{\delta}{2}$$

(ii) The momentum thickness

$$\theta = \int_{o}^{\delta} \frac{u}{U} \left( 1 - \frac{u}{U} \right) dy$$
$$= \int_{o}^{\delta} \frac{y}{\delta} \left( 1 - \frac{y}{\delta} \right) dy$$

$$= \int_{o}^{\delta} \left( \frac{y}{\delta} - \frac{y^2}{\delta^2} \right) dy$$

or.

$$\theta = \left[ \frac{y^2}{2\delta} - \frac{y^3}{3\delta^2} \right]_0^{\sigma} = \frac{\delta^2}{2\delta} - \frac{\delta^3}{3\delta^2} = \frac{\delta}{2} - \frac{\delta}{3} = \frac{\delta}{6}$$

iii. The energy thickness

$$\delta_{\varepsilon} = \int_{o}^{\delta} \frac{u}{U} \left( 1 - \frac{u^{2}}{U^{2}} \right) dy$$

$$= \int_{o}^{\delta} \frac{y}{\delta} \left( 1 - \frac{y^{2}}{\delta^{2}} \right) dy = \int_{o}^{\delta} \left( \frac{y}{\delta} - \frac{y^{3}}{\delta^{3}} \right) dy$$

$$= \left[ \frac{y^{2}}{2\delta} - \frac{y^{4}}{4\delta^{3}} \right]_{0}^{\delta} = \frac{\delta^{2}}{2\delta} - \frac{\delta^{4}}{4\delta^{3}}$$

$$= \frac{\delta}{2} - \frac{\delta}{4}$$

$$= \frac{\delta}{4}$$

iv. The value of  $\delta */\theta$ .

$$\frac{\delta^*}{\theta} = \frac{\delta/2}{\delta/6}$$

PROBLEM 2: The velocity distribution in the boundary layer is given by,

$$\frac{u}{U} = \frac{3}{2} \frac{y}{\sigma} - \frac{1}{2} \frac{y^2}{\sigma^2}$$

σ being the boundary layer thickness. Calculate the following

- (i) The ratio of displacement thickness to boundary layer thickness  $\left(\frac{\delta^*}{\delta}\right)$
- (ii) The ratio of momentum thickness to boundary layer thickness  $\left(\frac{\theta}{\delta}\right)$

## **Solution**

Velocity distribution: 
$$\frac{u}{U} = \frac{3}{2} \frac{y}{\sigma} - \frac{1}{2} \frac{y^2}{\sigma^2}$$

(i) 
$$\frac{\delta^*}{\delta}$$
:

$$\delta^* = \int_0^{\delta} \left( 1 - \frac{u}{U} \right) dy = \int_0^{\delta} \left( 1 - \frac{3}{2} \frac{y}{\sigma} + \frac{1}{2} \frac{y^2}{\sigma^2} \right) dy$$
$$= \left[ y - \frac{3}{2} \times \frac{y^2}{3\sigma} + \frac{1}{2} \times \frac{y^3}{3\sigma^2} \right]_0^{\delta}$$
$$\left[ \sigma - \frac{3}{4} \frac{\sigma^2}{\sigma} + \frac{1}{2} \frac{\sigma^2}{3\sigma^2} \right]$$

$$= \left(\sigma - \frac{3}{4}\sigma + \frac{\sigma}{6}\right)$$

$$\sigma^* = \frac{5}{12}\sigma$$
$$\therefore \frac{\sigma^*}{\sigma} = \frac{5}{12}\sigma.$$

(ii) 
$$\theta_{\sigma}$$

$$\theta = \int_0^\sigma \frac{u}{U} \left( 1 - \frac{u}{U} \right) dy$$

$$= \int_0^\sigma \left( \frac{3}{2} \frac{y}{\sigma} - \frac{1}{2} \frac{y^2}{\sigma^2} \right) \left( 1 - \frac{3}{2} \frac{y}{\sigma} + \frac{1}{2} \frac{y^2}{\sigma^2} \right) dy$$

$$= \int_0^\sigma \left( \frac{3}{2} \frac{y}{\sigma} - \frac{9}{4} \frac{y^2}{\sigma^2} + \frac{3}{4} \frac{y^3}{\sigma^3} - \frac{1}{2} \frac{y^2}{\sigma^2} + \frac{3}{4} \frac{y^3}{\sigma^3} - \frac{1}{4} \frac{y^4}{\sigma^4} \right) dy$$

$$= \int_0^\sigma \frac{3}{2} \frac{y}{\sigma} - \left( \frac{9}{4} \frac{y^2}{\sigma^2} + \frac{1}{2} \frac{y^2}{\sigma^2} \right) + \left( \frac{3}{4} \frac{y^3}{\sigma^3} + \frac{3}{4} \frac{y^3}{\sigma^3} \right) - \frac{1}{4} \frac{y^4}{\sigma^4} dy$$

$$= \int_0^\sigma \left( \frac{3}{2} \frac{y}{\sigma} - \frac{11}{4} \frac{y^2}{\sigma^2} + \frac{3}{4} \frac{y^3}{\sigma^3} - \frac{1}{4} \frac{y^4}{\sigma^4} dy \right) dy$$

$$= \left[ \frac{3}{2} \frac{y^2}{2\sigma} - \frac{11}{4} \frac{y^3}{3\sigma^2} + \frac{3}{2} \times \frac{y^4}{4\sigma^3} - \frac{1}{4} \times \frac{y^5}{4\sigma^4} \right]_0^\sigma$$

$$= \left[ \frac{3}{2} \times \frac{y^2}{2\sigma} \times \frac{11}{4} \times \frac{y^3}{3\sigma^2} + \frac{3}{2} \times \frac{\sigma^4}{4\sigma^3} - \frac{1}{4} \times \frac{\sigma^5}{5\sigma^4} \right]_0^{\delta}$$

$$\theta = \left( \frac{3}{4}\sigma - \frac{11}{12}\sigma + \frac{3}{8}\sigma - \frac{1}{20}\sigma \right) = \frac{19}{120}\sigma$$

**PROBLEM 3:** Find the displacement thickness, the momentum thickness and energy thickness for the velocity distribution in the boundary layer is given by

$$\frac{u}{U} = 2\left(\frac{y}{\delta}\right) - \left(\frac{y}{\delta}\right)^2.$$

Solution. Given:

Velocity distribution

$$\frac{u}{U} = 2\left(\frac{y}{\delta}\right) - \left(\frac{y}{\delta}\right)^2$$

(i) Displacement thickness  $\delta^*$  is given by equation

$$\delta^* = \int_0^\delta \left( 1 - \frac{u}{U} \right) dy$$

Substituting the value of

$$\frac{u}{U} = 2\left(\frac{y}{\delta}\right) - \left(\frac{y}{\delta}\right)^2$$
, we have

$$\delta^* = \int_0^{\delta} \left\{ 1 - \left[ 2 \left( \frac{y}{\delta} \right) - \left( \frac{y}{\delta} \right)^2 \right] \right\} dy$$

$$= \int_0^{\delta} \left\{ 1 - 2\left(\frac{y}{\delta}\right) + \left(\frac{y}{\delta}\right)^2 \right\} dy = \left[ y - \frac{2y^2}{2\delta} + \frac{y^3}{3\delta^2} \right]_0^{\delta}$$
$$= \delta - \frac{\delta^2}{\delta} + \frac{\delta^3}{3\delta^2} = \delta - \delta + \frac{\delta}{3} = \frac{\delta}{3}. \quad \text{Ans.}$$

(ii) Momentum thickness  $\theta$ , is given by equation

$$\theta = \int_0^{\delta} \frac{u}{U} \left\{ 1 - \frac{u}{U} \right\} dy = \int_0^{\delta} \left( \frac{2y}{\delta} - \frac{y^2}{\delta^2} \right) \left[ 1 - \left( \frac{2y}{\delta} - \frac{y^2}{\delta^2} \right) \right] dy$$

$$= \int_0^{\delta} \left[ \frac{2y}{\delta} - \frac{y^2}{\delta^2} \right] \left[ 1 - \frac{2y}{\delta} + \frac{y^2}{\delta^2} \right] dy$$

$$= \int_0^{\delta} \left[ \frac{2y}{\delta} - \frac{4y^2}{\delta^2} + \frac{2y^3}{\delta^3} - \frac{y^2}{\delta^2} + \frac{2y^3}{\delta^3} - \frac{y^4}{\delta^4} \right] dy$$

$$= \int_0^{\delta} \left[ \frac{2y}{\delta} - \frac{5y^2}{\delta^2} + \frac{4y^3}{\delta^3} - \frac{y^4}{\delta^4} \right] dy = \left[ \frac{2y^2}{2\delta} - \frac{5y^3}{3\delta^2} + \frac{4y^4}{4\delta^3} - \frac{y^5}{5\delta^4} \right]_0^{\delta}$$

$$= \left[ \frac{\delta^2}{\delta} - \frac{5\delta^3}{3\delta^2} + \frac{\delta^4}{\delta^3} - \frac{\delta^5}{5\delta^4} \right] = \delta - \frac{5\delta}{3} + \delta - \frac{\delta}{5}$$

$$= \frac{15\delta - 25\delta + 15\delta - 3\delta}{15} = \frac{30\delta - 28\delta}{15} = \frac{2\delta}{15} . \text{ Ans.}$$

(iii) Energy thickness  $\delta^{**}$  is given by equation

$$\begin{split} \delta^{**} &= \int_0^\delta \frac{u}{U} \bigg[ 1 - \frac{u^2}{U^2} \bigg] dy = \int_0^\delta \bigg( \frac{2y}{\delta} - \frac{y^2}{\delta^2} \bigg) \bigg( 1 - \bigg[ \frac{2y}{\delta} - \frac{y^2}{\delta^2} \bigg]^2 \bigg) dy \\ &= \int_0^\delta \bigg( \frac{2y}{\delta} - \frac{y^2}{\delta^2} \bigg) \bigg( 1 - \bigg[ \frac{4y^2}{\delta^2} + \frac{y^4}{\delta^4} - \frac{4y^3}{\delta^3} \bigg] \bigg) dy \\ &= \int_0^\delta \bigg( \frac{2y}{\delta} - \frac{y^2}{\delta^2} \bigg) \bigg( 1 - \frac{4y^2}{\delta^2} - \frac{y^4}{\delta^4} + \frac{4y^3}{\delta^3} \bigg) dy \\ &= \int_0^\delta \bigg( \frac{2y}{\delta} - \frac{y^2}{\delta^2} \bigg) \bigg( 1 - \frac{4y^2}{\delta^2} - \frac{y^4}{\delta^4} + \frac{4y^3}{\delta^3} \bigg) dy \\ &= \int_0^\delta \bigg( \frac{2y}{\delta} - \frac{8y^3}{\delta^3} - \frac{2y^5}{\delta^5} + \frac{8y^4}{\delta^4} - \frac{y^2}{\delta^2} + \frac{4y^4}{\delta^4} + \frac{y^6}{\delta^6} - \frac{4y^5}{\delta^5} \bigg) dy \\ &= \int_0^\delta \bigg[ \frac{2y}{\delta} - \frac{y^2}{\delta^2} - \frac{8y^3}{\delta^3} + \frac{12y^4}{\delta^4} - \frac{6y^5}{\delta^5} + \frac{y^6}{\delta^6} \bigg] dy \\ &= \bigg[ \frac{2y^2}{2\delta} - \frac{y^3}{3\delta^2} - \frac{8y^4}{4\delta^3} + \frac{12y^5}{5\delta^4} - \frac{6y^6}{6\delta^5} + \frac{y^7}{7\delta^6} \bigg]_0^\delta \\ &= \frac{\delta^2}{\delta} - \frac{\delta^3}{3\delta^2} - \frac{2\delta^4}{\delta^3} + \frac{12\delta^5}{5\delta^4} - \frac{\delta^6}{\delta^5} + \frac{\delta^7}{7\delta^6} = \delta - \frac{\delta}{3} - 2\delta + \frac{12}{5}\delta - \delta + \frac{\delta}{7} \\ &= -2\delta - \frac{\delta}{3} + \frac{12}{5}\delta + \frac{\delta}{7} = \frac{-210\delta - 35\delta + 252\delta + 15\delta}{105} \\ &= \frac{-245\delta + 267\delta}{105} = \frac{22\delta}{105}. \quad \text{Ans.} \end{split}$$

PROBLEM 4: For the velocity for the laminar boundary layer flows given as

$$\frac{u}{U} = 2\left(\frac{y}{\sigma}\right) - \left(\frac{y}{\sigma}\right)^2$$

find out the expression for boundary layer thickness ( $\delta$ ), shear stress ( $\tau_0$ ), co-efficient of drag ( $C_D$ ) in terms of Reynolds number.

Solution. Given:

(i) The velocity distribution 
$$\frac{u}{U} = 2\left(\frac{y}{\delta}\right) - \left(\frac{y}{\delta}\right)^2$$
 ...(i)

Substituting this value of  $\frac{u}{U}$ , we get

$$\frac{\tau_o}{\rho U^2} = \frac{d}{dx} \left[ \int_0^{\sigma} \left( \frac{2y}{\sigma} - \frac{y^2}{\sigma^2} \right) \left( 1 - \left( \frac{2y}{\sigma} - \frac{y^2}{\sigma^2} \right) \right) dy \right]$$

$$= \frac{d}{dx} \left[ \int_0^{\sigma} \left( \frac{2y}{\sigma} - \frac{y^2}{\sigma^2} \right) \left( 1 - \frac{2y}{\sigma} + \frac{y^2}{\sigma^2} \right) dy \right]$$

$$= \frac{d}{dx} \left[ \int_0^{\sigma} \left( \frac{2y}{\sigma} - \frac{4y^2}{\sigma^2} + \frac{2y^3}{\sigma^3} - \frac{y^2}{\sigma^2} + \frac{2y^3}{\sigma^3} - \frac{y^4}{\sigma^2} \right) dy \right]$$

$$= \frac{d}{dx} \left[ \int_0^{\sigma} \left( \frac{2y}{\sigma} - \frac{5y^2}{\sigma^2} + \frac{4y^3}{\sigma^3} - \frac{y^4}{\sigma^4} \right) dy \right]$$

$$= \frac{d}{dx} \left[ \frac{2}{2} \frac{y^2}{\sigma} - \frac{5}{3} \frac{y^2}{\sigma^2} + \frac{4}{4} \frac{y^4}{\sigma^3} - \frac{1}{5} \frac{y^5}{\sigma^4} \right]_0^{\sigma}$$

$$= \frac{d}{dx} \left[ \sigma - \frac{5}{3} \sigma + \sigma \frac{1}{5} \sigma \right] = \frac{d}{dx} \left( \frac{2}{15} \sigma \right)$$

$$\therefore \tau_o = \rho U^2 \times \frac{d}{dx} \left( \frac{2}{15} \sigma \right) = \frac{2}{15} \rho U^2 \frac{d\delta}{dx} - \dots (ii)$$

Also, according to Newton's law of viscosity

$$\tau_o = \mu \left(\frac{dy}{dx}\right)_{y=0} - - - - - - - - (iii)$$

$$But \quad u = U\left(\frac{2y}{\sigma} - \frac{y^2}{\sigma^2}\right)$$

$$But \quad u = U\left(\frac{2y}{\sigma} - \frac{y^2}{\sigma^2}\right)$$

$$and \quad \frac{du}{dx} = U\left(\frac{2}{\sigma} - \frac{2y}{\sigma^2}\right), U \text{ being constan } t$$

$$\therefore \left(\frac{du}{dx}\right) = U\left(\frac{2}{\sigma} - 0\right) = \frac{2U}{8}$$

Substituting this value in (iii), we get

$$\tau_o = \frac{2\mu U}{\sigma} - - - - - (iv)$$

Equating the values of  $\tau_o$  given by equations (ii) and iv, we get

$$\frac{2}{15}\rho U^2 \frac{d\sigma}{dx} = \frac{2\mu U}{\sigma}$$

or 
$$\sigma \cdot \frac{d\sigma}{dx} = \frac{15\mu U}{\rho U^2} = \frac{15\mu}{\rho U^2}$$
  
or  $\sigma \cdot d\sigma = \frac{15\mu}{\rho U} dx$ 

Integrating both sides, we get

$$\frac{\delta^2}{2} = \frac{15\mu}{\rho U}x + c \quad (where \ C = Cons \tan t \ of \ integration)$$

At 
$$x = 0$$
,  $\delta = 0$ :  $C = 0$ 

$$\therefore \frac{\delta^2}{2} = \frac{15\mu}{\rho U}$$

or 
$$\delta = \sqrt{\frac{2 \times 15 \mu x}{\rho U}} = 5.48 \sqrt{\frac{\mu x}{\rho U}}$$
  
=  $5.48 \sqrt{\frac{\mu x \times x}{\rho U \times x}} = 5.48 \sqrt{\frac{x^2}{Re_x}}$ 

(where 
$$\operatorname{Re}_{x} = \frac{\rho Ux}{\varpi}$$
,)  
or  $\sigma = 5.48 = \frac{x}{\sqrt{\operatorname{Re}_{x}}} - - - - - (v)$ 

(ii) Shear stress  $\tau_o$ :

From equation (iv), we have

$$\tau_o = \frac{2\mu U}{\sigma}$$

But 
$$\sigma = 5.48 \frac{x}{\sqrt{\text{Re}_x}}$$

$$\therefore \tau_o = \frac{2\mu U}{5.48 \frac{x}{\sqrt{\text{Re}_x}}} = \frac{2\mu U \sqrt{\text{Re}_x}}{5.48 x} = 0.365 \frac{\mu U}{x} \sqrt{\text{Re}_x} - - - - - - - - - (vi)$$

(iii) Local Co-efficient of drag,  $C_D^*$ 

$$\tau_o = \frac{0.365 \,\mu U}{x} = \sqrt{\text{Re}_x}$$
 also 
$$\tau_o = C_D^* \frac{\rho U^2}{2} - - - - - - - - (vii) \big( \text{where } C_D^* = \text{local coefficient of drag} \big)$$

Equating the two of  $\tau_o$ , given by equation (vi) and (vii), we get

$$C_D^* = \frac{0.365 \mu U}{x} \sqrt{\text{Re}_x} \text{ or } C_D^* = 0.365 \times 2 \times \frac{\sqrt{\text{Re}_x}}{\rho U x}$$

$$= \frac{0.73}{\sqrt{\text{Re}_x}}$$

(iv) Co-efficient of drag, CD:

We know that 
$$C_D = \frac{F_D}{\frac{1}{2} \rho A U^2}$$

Where, 
$$F_D = \int_0^L \tau_o \times B \times dx$$

$$= \int_0^l 0.365 \, \frac{\mu U}{x} \, \sqrt{\text{Re}_x} \, \times B \times \, dx$$

$$= 0.365 \int_0^l \frac{\mu U}{x} \sqrt{\frac{\rho U x}{\mu}} \times B \times dx \left( \because \text{Re}_x = \frac{\rho U x}{\mu} \right)$$

$$= 0.365 \int_0^1 \frac{\mu U}{x} \sqrt{\frac{\rho U x}{\mu}} \times \frac{1}{\sqrt{x}} \times B \times dx$$

$$= 0.365 \ \mu U \ \sqrt{\frac{\rho U}{\mu}} \times B \int_0^l x^{-\frac{1}{2}} \times dx$$

$$=0.365 \ \mu U \ \sqrt{\frac{\rho U}{\mu}} \times B \left[ \frac{x^{-\frac{1}{2}}}{\frac{1}{2}} \right]$$

= 
$$0.365 \times 2 \mu UB \sqrt{\frac{\rho U}{\mu}} \times B \sqrt{L}$$

$$\therefore C_D = \frac{0.73 \,\mu U B \sqrt{\frac{\rho U}{\mu}}}{\frac{1}{2} \,\rho A U^2}$$

(Where A - area of plate = L x B, L and B being length and width of the plate respectively)

$$\therefore C_D = \frac{0.73 \mu U B \sqrt{\frac{\ell U L}{\mu}}}{\frac{1}{2} \ell \times L \times B \times U^2} = \frac{1.46 \mu}{\ell L.U} \sqrt{\frac{\ell U L}{\mu}}$$
$$= \frac{1.46 \sqrt{\mu}}{\sqrt{\ell L.U}} = 1.46 \sqrt{\frac{\mu}{\ell L.U}} = \frac{1.46}{\sqrt{\text{Re}_l}}$$



OBSERVE OPTIMIZE OUTSPREAD